

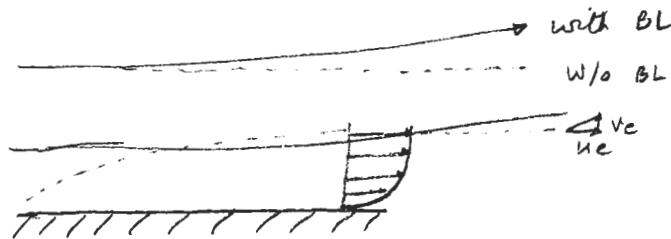
A) 2D Inviscid Models

B) Implication for Lift and Drag

Reading: Handout, paper

- A) Zeroth order matching $\vec{u} \cdot \hat{n} = 0$
- First order matching $\vec{u} \cdot \hat{n} = V_e = V_i \quad (\sim O(1/\sqrt{Re}))$

In actual flow

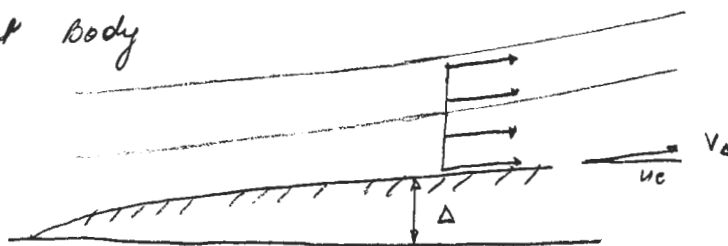


Using continuity $\nabla \cdot \vec{u} = 0$ gives

$$V_0 = \frac{d}{dx} (u_e \delta^*) - \delta \frac{du_e}{dx}$$

Two models to enforce V_e :

① Displacement body



$$\int_0^{y_e} () = \int_0^{\Delta} () + \int_{\Delta}^{y_e} ()$$

$$\Rightarrow V_e - V_0 = - \int_{\Delta}^{y_e} \frac{\partial u}{\partial x} dy = \frac{d(\Delta u_e)}{dx} - y_e \frac{du_e}{dx} - u_e \frac{dy_e}{dx} + u_e \frac{dy_e}{dx} - u_e \frac{d\Delta}{dx}$$

$$V_e - V_0 = \frac{d}{dx}(\Delta u_e) - y_e \frac{du_e}{dx} - u_e \frac{d\Delta}{dx}$$

$$V_0 = u_e \frac{d\Delta}{dx} \quad (\text{by observation})$$

$$\therefore V_e = \frac{d}{dx}(\Delta u_e) - y_e \frac{du_e}{dx}$$

$\Rightarrow \Delta = \delta^*$ on comparing with real visc. flow

Note $\vec{u}_{EIF} \cdot \hat{n}_{imp} = 0$

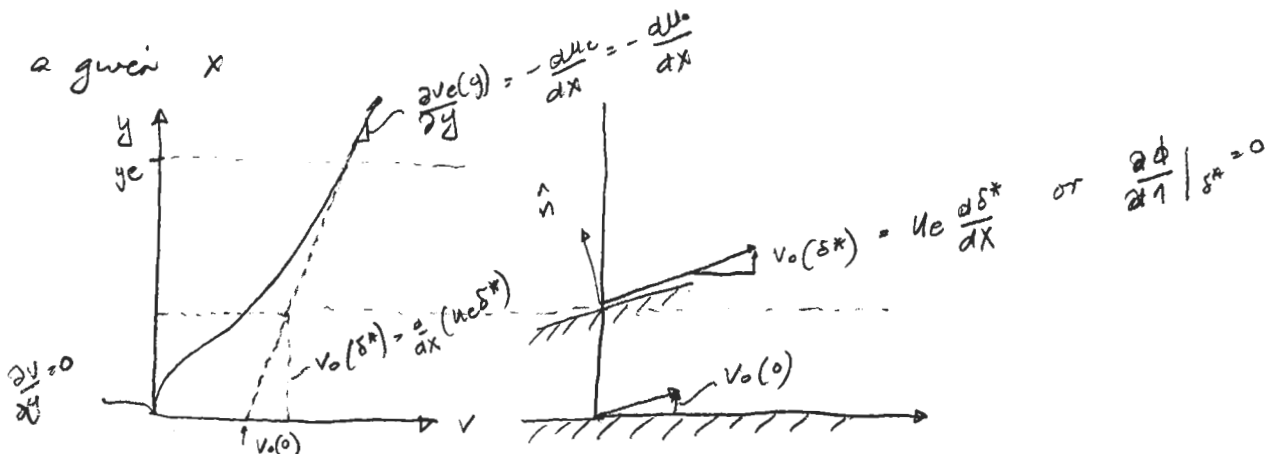
② Wall blowing

$$\begin{aligned} V_e - V_{wall} &= - \int_0^{y_e} \frac{\partial u}{\partial x} dy \\ &= - \frac{d}{dx} \int_0^{y_e} u dy + u_e \frac{dy_e}{dx} \\ &= - \frac{d}{dx} u_e y_e + u_e \frac{dy_e}{dx} \\ &= - y_e \frac{du_e}{dx} \end{aligned}$$

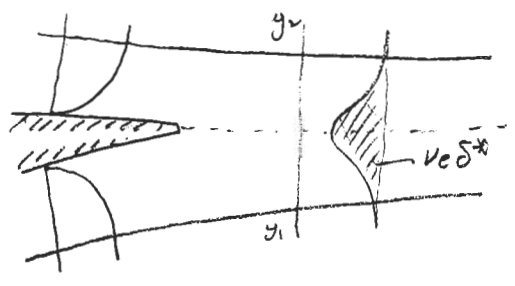
$$\therefore V_e = V_{wall} - y_e \frac{du_e}{dx}$$

$$\Rightarrow V_{wall} = \frac{d}{dx} (u_e \delta^*)$$

At a given x



Wake



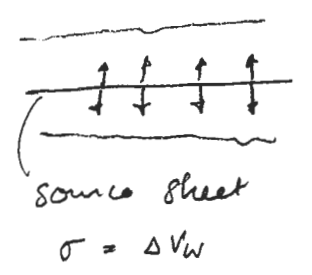
Real flow:
$$V_2 - V_1 = - \int_{y_1}^{y_2} \frac{\partial u}{\partial x} = \frac{d}{dx} (u_e \delta^*) - (y_2 - y_1) \frac{du_e}{dx}$$

Disp. Body
$$V_2 - V_1 = \frac{d}{dx} (u_e \Delta) - (y_2 - y_1) \frac{du_e}{dx} \Rightarrow \Delta = \delta^*$$

Wall Blowing

$$V_2 - V_1 = V_{w2} - V_{w1} - (y_2 - y_1) \frac{du_e}{dx}$$

$$\Rightarrow V_{w2} - V_{w1} = \frac{d}{dx} (u_e \delta^*)$$



- No info on where to put body.
- No info on individual v_{w1}, v_{w2} ($\Delta \left(\frac{\partial \phi}{\partial \eta} \right) = \Delta v_w$)
- Wake portion at $\Delta p = 0$ ($\Delta \left(\frac{\partial \phi}{\partial s} \right) = 0$)

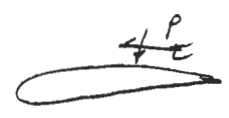
potential flow controls difference in u_e
 b.c " difference in v_w

20 Drag (implication of FDLT)

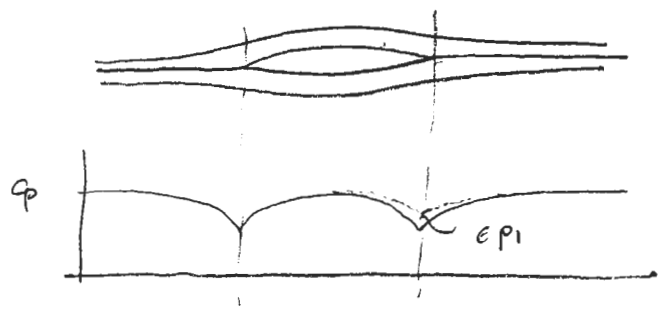
$$D = \int -p dy + \int \tau dx$$

\uparrow
 D_{form} or D_{press}

\uparrow
 Friction



For inviscid, stress free, flow $D_{form} \equiv 0$ (d'Alembert's Paradox)
 perfect cancellation



For classical theory, wall BC: $v_0 = 0 \rightarrow$ flow tangency

$$D_{form} = 0$$

For IBLT, wall BC: $v_0 = v_i \sim O(1/\sqrt{Re})$ attached
 $\sim O(1)$ separated

$$v_0 = u_{inv} + u_{vis}$$

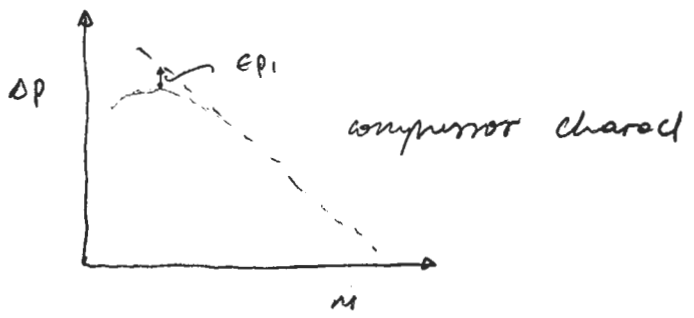
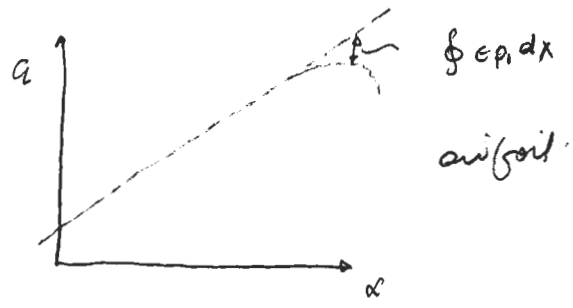
\uparrow \uparrow
 $O(1)$ $O(1/\sqrt{Re})$

\downarrow \downarrow

By $p = p_{inv} + p_{vis}$

$$D_{form} = 0 + \frac{O(1/\sqrt{Re})}{O(1)} \leftarrow \text{about } 1/2 \text{ the total drag.}$$

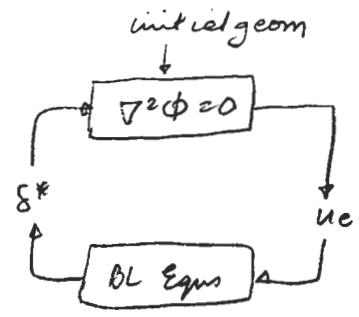
Similarly for lift (stall prediction)



5.3 IBLT

- A) Iteration Stability
- B) Fully Coupled Solution

A) Iteration Stability



Classical Iteration

- 0) assume some $\delta^* (=0)$
- 1) add δ^* to airfoil contour ($\frac{\partial \phi}{\partial y} = v_{wall}$ or)
- 2) solve $\nabla^2 \phi = 0 \rightarrow u_e = \frac{\partial \phi}{\partial x}$
- 3) solve BL eqn \rightarrow calc δ^*
- 4) iterate

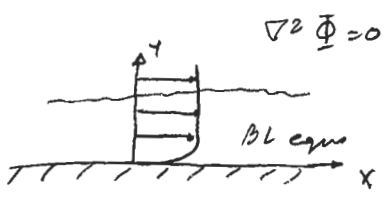
Problem: Almost never works due to numerical instability

stability Analysis

- Assume a converged solution
- Perturb solution
- Examine if perturbation decays with iteration $n \rightarrow \infty$

Consider BL flow over a wall

let $\bar{\Phi} = u_{\infty}(x + \phi)$
 ↑ perturbation potential



$$u = \bar{\Phi}_x = u_{\infty}(1 + \phi_x) \quad \frac{du}{dx} = u_{\infty} \phi_{xx} \quad (\nabla^2 \bar{\Phi} = \nabla^2 \phi = 0)$$

$$\frac{\partial \phi}{\partial y} = \alpha \quad \text{boundary conditions}$$

linearize BL eqns.

$$\begin{matrix} H \text{ [norm]} \\ \theta \text{ [k-E shape par]} \end{matrix} \Rightarrow \frac{d\delta^*}{dx} = A + B \frac{dUe}{dx}$$

\uparrow \uparrow
 C_0, C_0 H constant
 const

Assume A, B fixed and depend on base solution

Note that

$$\alpha = \frac{d\delta^*}{dx} \text{ — slope of displacement surface (no flow through surface)}$$