16.522: Session 18 Courtesy of Prof. Eduardo Ahedo Universidad Politecnica de Madrid

HALL THRUSTERS: FLUID MODEL OF THE DISCHARGE



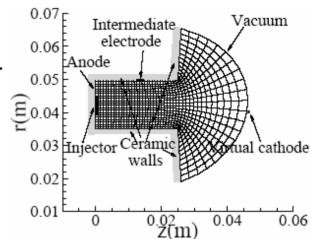
Types of models

- **Kinetic:** based on Boltzmann eq.; unaffordable except for particular aspects of the problem
- Fluid: familiar formulation but important difficulties arising from
 - 1. Weak collisionality (Kn large)
 - 2. Wall interaction
 - 3. Curved magnetic topology
 - 4. 2D subsonic/supersonic ion flows
- Particle In Cell & MonteCarlo methods (PIC/MCC): good for weak collisionality; simple to implement, but subject to 'numerical effects'; important difficulties in dealing with disparate scales of electron and ions. MCC model collisions statistically.
- Hybrid (PIC/MCC for heavy species & fluid for electrons): best compromise today; allows 2D (geometrical and magnetic) effects; avoids small electron scales and admits quasineutrality

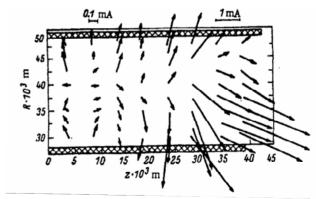


Simulation of the plasma discharge

- 2D, axisymmetric model
- Quasineutral plasma except for sheaths around walls.
- Plasma wall interaction treated in separate sheath models.
- Boundaries: 1) anode + gas injector, 2) cathode surface,
 3) lateral walls
- 3 or 4 species: neutrals, ions (+, ++), and electrons
- Ion dynamics: unmagnetized, near collisionless, internal regular sonic transitions, singular sonic transitions at sheath edges.
- Electrons: magnetized, diffusive motion, and weakly collisional (local thermodynamic equilibrium is not assured)
- Fluid modelling: complex and uncertain.
- There is no fully 2D model yet. Two existing options:
 - a) Approximate 1D (axial) fluid model
 - b) Near 2D hybrid model: fluid eqs. for electrons; particle model for ions & neutrals © America



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2D fluid model

- Basic hypotheses: 1) azimuthal symmetry: $\frac{\partial}{\partial \theta} = 0$, $u_{\theta i} = 0$, $u_{\theta n} = 0$
 - 2) Quasineutrality: $|n_e = n_{i+} + 2n_{i++}| \Rightarrow$ boundary conditions at sheath edges
 - 3) Simplified treatment of pressure tensors

Continuity equations:
$$\frac{\partial n_e}{\partial t} + \nabla \cdot n_e \vec{u}_e = S_{ion}$$
 $S_{ion} \approx n_e n_n R_{ion}(T_e) \leftarrow \text{ionization}$

$$S_{ion} \approx n_e n_n R_{ion}(T_e) \leftarrow \text{ionization}$$

$$\left[\frac{\partial n_i}{\partial t} + \nabla \cdot n_i \vec{u}_i = S_{ion}, \quad \frac{\partial n_n}{\partial t} + \nabla \cdot n_n \vec{u}_n = -S_{ion} \right] \quad \text{(here } n_e = n_i \text{ only)}$$

(here
$$n_e = n_i$$
 only)

Electron momentum:
$$\left| \frac{\partial}{\partial t} (m_e n_e \vec{u}_e) + \nabla \cdot m_e n_e \vec{u}_e \vec{u}_e \right| = -\nabla (n_e T_e) - e n_e (\vec{E} + \vec{u}_e \times \vec{B}) + \vec{M}_e$$

$$\vec{M}_{e} \approx -n_{e}n_{n}R_{en}(T_{e})m_{e}\vec{u}_{e} \leftarrow \text{e-n momentum transfer}$$

• Ion momentum:
$$\frac{\partial}{\partial t} (m_i n_i \vec{u}_i) + \nabla \cdot m_i n_i \vec{u}_i \vec{u}_i = -\nabla (n_i T_i) + e n_i \vec{E} + \vec{M}_i$$

$$\vec{M}_i \approx S_{ion} m_i \vec{u}_n - S_{cx} m_i (\vec{u}_i - \vec{u}_n) \leftarrow \text{ionization} + \text{charge-exchange mom. transf.}$$

2D fluid model

• Electron (total) energy equation :

$$\boxed{\frac{\partial}{\partial t} \left(\frac{3}{2} T_e + \frac{1}{2} m_e u_e^2 \right) n_e + \nabla \cdot \left[\left(\frac{5}{2} T_e + \frac{1}{2} m_e u_e^2 \right) n_e \vec{u}_e + \vec{q}_e \right] = -e n_e \vec{u}_e \cdot \vec{E} + Q_e}$$

$$Q_e \approx -n_e n_n R_{ion}(T_e) E_{ion} \alpha_{ion} \leftarrow \text{ionization} \quad (E_{ion} = 12.1 \text{eV}, \quad \alpha_{ion} \sim 1.5 - 2.5 \text{ for Xe})$$

• Internal energy = total energy – mechanical energy

$$\left[\frac{\partial}{\partial t} \left(\frac{3}{2} T_e n_e\right) + \nabla \cdot \left[\frac{3}{2} T_e n_e \vec{u}_e + \vec{q}_e\right] = -n_e T_e \nabla \cdot \vec{u}_e + Q'_e$$

$$Q'_{e} \equiv Q_{e} - \vec{M}_{e} \cdot \vec{u}_{e} + \frac{1}{2} m_{e} u_{e}^{2} S_{e} \approx n_{e} n_{n} \left[\left(R_{en} + \frac{1}{2} R_{ion} \right) m_{e} u_{e}^{2} - R_{ion} E_{ion} \alpha_{ion} \right]$$

• Heat flux (diffusive model): (Bittencourt)

$$\left| \vec{o} = -\frac{5}{2} p_e \nabla T_e - e \vec{q}_e \wedge \vec{B} - m_e v_e \vec{q}_e \right| \rightarrow \vec{q}_e = -\overline{\vec{K}}_e \cdot \nabla T_e$$

• Energy equations for ions and neutrals.



From 2D to 1D model

Electron continuity equation:

$$\frac{\partial n_e}{\partial t} + \frac{\partial n_e u_{ze}}{\partial z} + \frac{1}{r} \frac{\partial r n_e u_{re}}{\partial r} = n_e v_i$$

The 1D axial model works with values averaged over each ('doughnut') radial section

$$0 = \frac{2}{r_{\text{ext}}^2 - r_{\text{int}}^2} \int_{r_{\text{int}}}^{r_{\text{ext}}} \left[\frac{\partial n_e}{\partial t} + \frac{\partial n_e u_{ze}}{\partial z} + \frac{1}{r} \frac{\partial r n_e u_{re}}{\partial r} - n_e v_i \right] \rightarrow \left[\frac{\partial \overline{n}_e}{\partial t} + \frac{\partial \overline{n}_e \overline{u}_{ze}}{\partial z} = \overline{n}_e (\overline{v}_i - \overline{v}_w) \right]$$

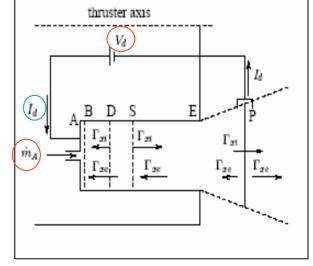
with
$$\overline{n}_e \equiv \overline{n}_e(z)$$
,..., and 'source' $\boxed{\overline{n}_e \overline{v}_w = 2 \frac{(r n_e u_{re})|_{\text{int}}^{ext}}{r_{ext}^2 - r_{\text{int}}^2}}$ evaluating losses at lateral walls.

An auxiliary radial model (at each z) is needed to

compute \overline{V}_{w} & determine radial profiles: $\left| \frac{1}{r} \frac{\partial r n_{e} u_{re}}{\partial r} \approx n_{e} \overline{V}_{w} \right|$

$$\frac{1}{r} \frac{\partial r n_e u_{re}}{\partial r} \approx n_e \overline{V}_w$$

- This is a variable-separation type of solution; $\overline{V}_{w}(z)$ is an eigenvalue of the radial model.
- We proceed similarly with the rest of fluid equations.





1D axial model

- Neglect jet divergence, doubly-charged ions, $p_i, p_n,...$
- Wall interaction terms appear as source terms (instead of BCs)
- Conservation of species flows:

$$\left| \frac{d}{dz} (n_e u_{zi}) = n_e (v_i - v_w) \right|$$

 v_i : ionization frequency v_w : recombination frequency

$$n_n u_{zn} + n_e u_{zi} = \text{const} = \dot{m}_A / A m_i$$

$$n_e u_{zi} - n_e u_{ze} = \text{const} = I_d / Ae$$

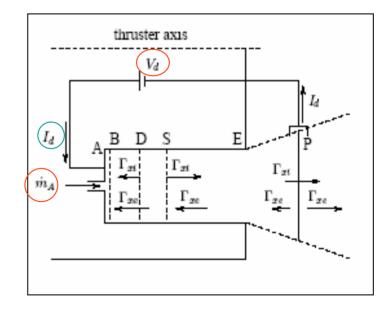
• Ion axial momentum equation:

$$\frac{d}{dz}(m_{i}n_{e}u_{zi}^{2}) = -en_{e}\frac{d\phi}{dz} + m_{i}n_{e}(v_{i}u_{zn} - v_{w}u_{zi})$$

Neutral axial momentum equation

$$\left| \frac{d}{dz} (m_i n_n u_{zn}^2) = m_i n_e (v_{wn} u_{znw} - v_i u_{zn}) \right|$$

 u_{znw} velocity of neutrals from recombination $(\neq u_{zi})$





1D axial model

$$u_{\theta e} = -u_{ze} \frac{\omega_e}{v_e}$$

 $v_e = v_{en} + v_{ei} + v_{wm} + v_{turb}$: effective collision frequency (v_{wm} due to wall interaction, v_{turb} due to plasma turbulence)

Electron internal energy:

$$\boxed{\frac{d}{dz}\left(\frac{5}{2}n_eT_eu_{ze} + q_{ze}\right) = u_{ze}\frac{d(n_eT_e)}{dz} + v_en_em_eu_e^2 - v_in_eE_{ion}\alpha_{ion} - \beta_ev_wn_eT_e}$$

 $(\beta_e \sim 6-100$: factor for energy losses at lateral walls)

• Heat conduction:
$$0 = -\frac{5}{2} p_e \nabla T_e - e \vec{q}_e \wedge \vec{B} - m_e v_e \vec{q}_e \rightarrow q_{ze} = -\frac{5n_e T_e}{2m_e} \frac{v_e}{\omega_e^2} \frac{dT_e}{dz}$$

Normalized set of equations: $\left| (T_e - m_i u_{zi}^2) \frac{dY}{dz} = \vec{F}(\vec{Y}) \right|$

 $\vec{Y} = (n_{o}, n_{n}, u_{\tau i}, u_{\tau o}, u_{\tau n}, T_{o}, q_{\tau e}, \phi)$ vector of 8 plasma variables

Singular (sonic) points of the equations at: $M = \frac{u_{zi}}{\sqrt{T/m_z}} = \pm 1$



1D axial model

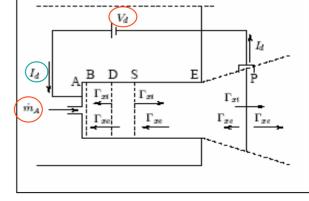
• For instance:
$$\frac{du_{zi}}{dz} = v_i - u_{zi} \frac{d \ln A}{dz} - u_{zi} \frac{G}{T_e - m_i u_{zi}^2}$$

$$G = -\frac{\omega_e^2}{v_e} m_e u_{ze} \left(1 - \frac{2q_{ze}}{5n_e T_e u_{ze}} \right) - v_i m_i (2u_{zi} - u_{zn}) + m_i u_{zi}^2 \frac{d \ln A}{dz}$$

- A singular transition exists at anode sheath edge (point B): $M_B = -1$ & $G_B = \infty$
- A regular subsonic/supersonic transition exists inside the channel (point S) with

$$M_S = 1 \& \vec{F}(\vec{Y}_S) = \vec{0}$$
, i.e. $G_S = 0$

- Point S is equivalent to the critical section of a Laval nozzle, where $0=G_S\equiv m_iu_{zi}^2d\ln A/dz$ corresponds to dA/dz=0. In a Hall thruster, $G_S=0$ corresponds to a certain balance between ionization and electron diffusion.
- 8 boundary conditions are set at points B, S and P:
 - Discharge voltage (1)
 - Density and velocity of injected gas at anode (2)
 - Electron temperature at cathode (1)
 - Regularity condition at point S (1)
 - Anode sheath conditions (3)





Anode sheath in a Hall thruster

- For Maxwellian-type VDF electron flux to anode $\sim n_{eA} \sqrt{T_e/2\pi m_e}$,
- In normal conditions this is much larger than the quasineutral flux in the channel, $g_{ze} \equiv n_e u_{ze} (>0)$
- A negative anode sheath AB is formed in order to

satisfy
$$g_{zeB} = g_{zeA}$$
:
$$n_{eB}u_{zeB} \approx -n_{eB} \exp\left(-\frac{e\phi_{AB}}{T_{eB}}\right) \sqrt{\frac{T_{eB}}{2\pi m_e}}$$

This equation determines the sheath potential fall, ϕ_{AB} .

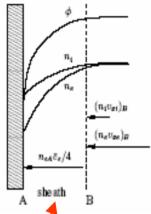
- Bohm condition at sheath edge: $u_{ziB} \approx -\sqrt{T_{eB}/m_i}$
- Electron energy flux deposited: a) at the anode A,

$$q_{zeA}^{tot} = -\int \frac{1}{2} m_e w^2 w_z f_{eB}(w) d^3 w \approx 2T_{eB} g_{zeB}$$

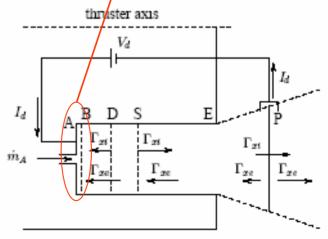
b) at the sheath edge B:

$$q_{zeB}^{tot} = g_{zeB}(2T_{eB} + e\phi_{AB})$$

• These are 3 boundary conditions for the axial quasineutral model



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Axial structure

Acceleration region:

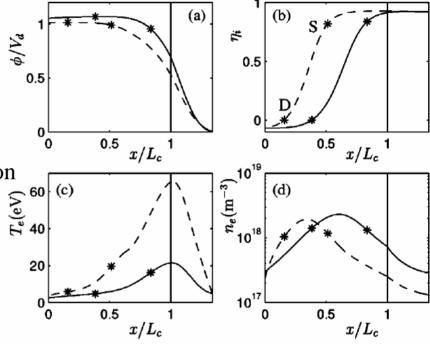
- Presents most of the potential drop & ion acceleration
- For electrons: Joule heating competes with wall cooling
- Heat conduction smooths T_e profile
- Plasma production = plasma recombination
- Plasma density decreases (due to ion acceleration)

Ionization region:

- Electron cooling due to ionization
- Maximum of plasma density inside

• Ion backstreaming region:

- Electric force very small
- Ion reverse flow is small
- Pressure drop (towards anode) dominates electron diffusive flow
- Length depends on ionization rate (i.e. T_e)



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• Magnetic field is adjusted for each case:

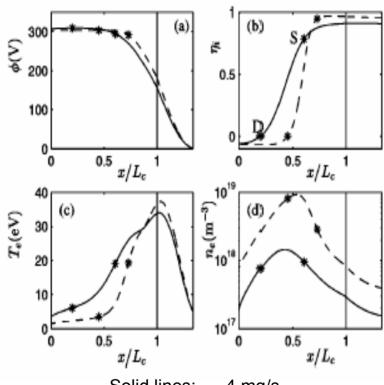
- solid lines: 110V, 110G

- dashed lines: 600V, 330G



Axial structure

Influence of anode mass flow



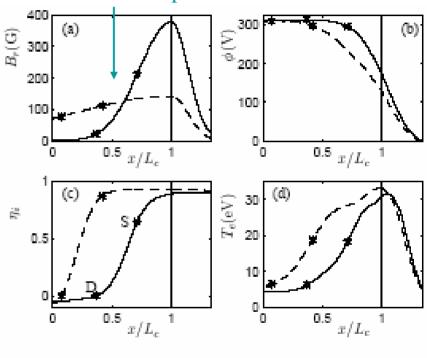
- Solid lines: 4 mg/s

- Dashed lines: 10 mg/s

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Influence of magnetic field shape

External B-field profiles



- solid lines: $L_{b1} = 9.5 \text{mm}$, $B_{max} \sim 380 \text{ G}$

- dashed lines: $L_{b1} = 30 \text{mm}$, $B_{max} \sim 140 \text{G}$

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Thrust

• Plasma momentum flow:
$$F_p(z) = \sum_{\alpha=i,e,n} F_{\alpha}(z)$$
, $F_{\alpha}(z) = (m_{\alpha}u_{z\alpha}^2 + T_{\alpha}) n_{\alpha}A$

• Axial momentum equation for the plasma:

$$\frac{dF_i}{dz} = Aen_i \frac{d\phi}{dz} + Am_i n_e (v_i u_{zn} - v_w u_{zi})$$

Electric force

$$\frac{dF_n}{dz} = Am_i n_e (v_w u_{znw} - v_i u_{zn})$$

 $\frac{dF_e}{dz} = Aen_e \frac{d\phi}{dz} + Aj_{\theta e}B_r + n_e T_e \frac{dA}{dz}$





Ionization + wall recombination

$$\Rightarrow \left(\frac{dF_p}{dz}\right) = \frac{\varepsilon_0}{2} \frac{d}{dz} \left(\frac{d\phi}{dz}\right)^2 + \left(Aj_{\theta e}B_r\right) Av_w m_i (u_{zi} - u_{znw}) n_e + n_e T_e \frac{dA}{dz}$$

Non-neutral

E-field force



Thrust

• Integrating the preceding equation between anode and far downstream:

• Thrust:
$$F = F_{p\infty} - D_{plume} = F_{mag} + F_{pA} - F_{elec} - D_{wall}$$

- Magnetic force:
$$F_{mag} = \int_{A}^{\infty} j_{\theta e} B_r A dz$$

- Electric force:
$$F_{elec} = \left[A \frac{\varepsilon_0}{2} \left(\frac{d\phi}{dz} \right)^2 \right]_A$$

- Ion wall impact drag:
$$D_{wall} = \int_{A}^{E} v_w m_i (u_{zi} - u_{znw}) n_e A dz$$

- Jet divergence drag:
$$D_{plume} = \int_{A}^{\infty} n_e T_e \frac{dA}{dz} dz$$

- Therefore, thrust is transmitted to the thruster through the magnetic reaction force of electrons on the thruster magnetic circuit.
 - Notice the contribution of the external magnetic field to thrust.
 - Pressure forces at the anode make a marginal contribution
 - Ion energy accommodation at walls acts as a drag force.



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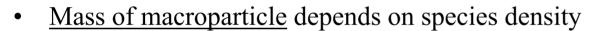


HALL THRUSTERS: HYBRID 2D CODE



Particle-in-cell (PIC) methods

- Individual motion of <u>macroparticles</u>, subject to <u>electromagnetic forces</u>, is followed in a <u>computational grid</u>. (see Birdsall-Langdon)
- <u>Cell size</u>, l_{cell} smaller than plasma gradients
- Timestep, $\Delta \tau$, such that particles advance less than cell size.
- Number of macroparticles per cell, N_{cell}, depends on good statistics and small numerical oscillations.



- Different species → different sets of macroparticles
- Example for Hall thruster (only ions and neutrals)
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Intermediate

electrode

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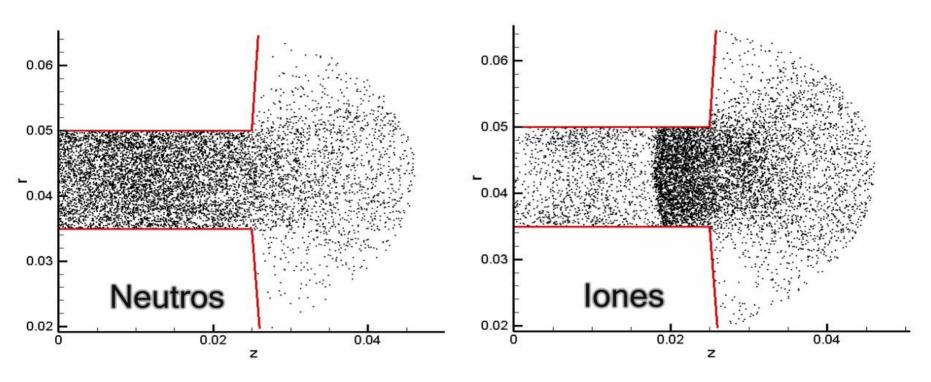
Vacuum

ual cathode

- axisymmetric, 30mm × 15 mm → 900 (toroidal) cells → l_{cell} ~0.5mm
- N_{cell} (per species) ~ 30 → 27.000 macroparticles/species → 10^9 - 10^{11} atoms/macroparticle
- macroparticle mass ~ atom mass × (atom/macroparticle)
- $-\Delta \tau \sim 10^{-8}$ s (ions), 10^{-7} s (neutrals), 10^{-10} s (electrons)



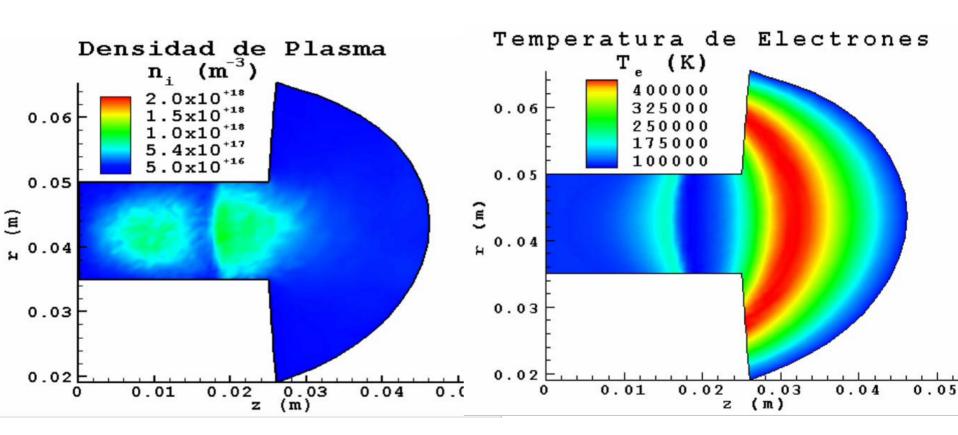
Video of particle motion



Courtesy of Professor Ahedo, Universidad Carlos III de Madrid. Used with permission.



Video of plasma dynamics



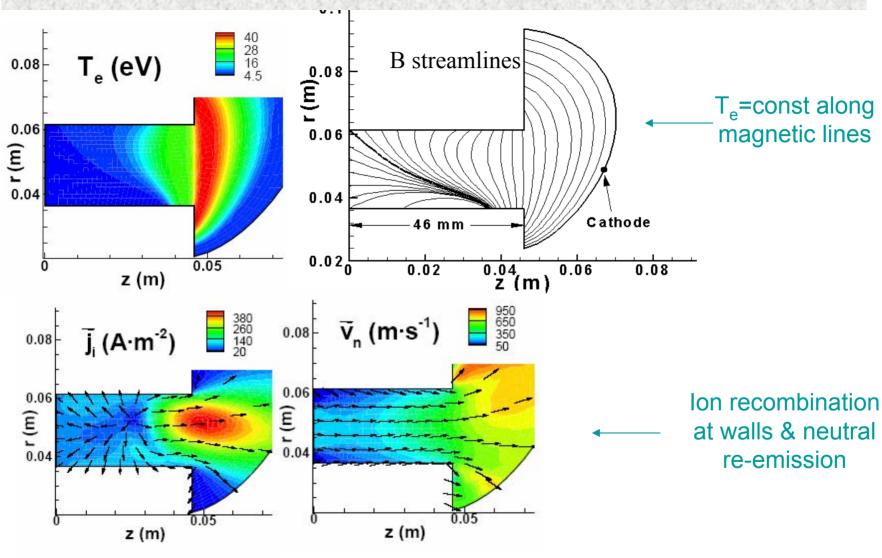
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Time-averaged 2D behavior

:][i fY`fYa cj YX`Xi Y`hc`Wtdnf][\h`fYghf]Wh]cbg":D`YUgY`gYY`:][i fY`%%`]b`9gWtVUfž`8"ž`5"`5bhŒbž`UbX`9"`5\YXc"
"G]a i `Uh]cb`cZ<][\!GdYW]Z]WV=a di `gY`UbX`8ci V`Y!GhU[Y`<U``H\fi ghYfg""`=b`DfcW*&-h\`=bhYfbUh]cbU``9`YWMf]W
Dfcdi `g]cb`7cbZYfYbWYž`Df]bWYhcbž'I G5"`&\$\$) "



Time-averaged 2D behavior



7ci fhYgmcZ'DfcZYggcf'5\YXcž'I b]j Yfg]XUX'7Uf'cg' — XY'A UXf]X"'I gYX'k]h\ 'dYfa]gg]cb"



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