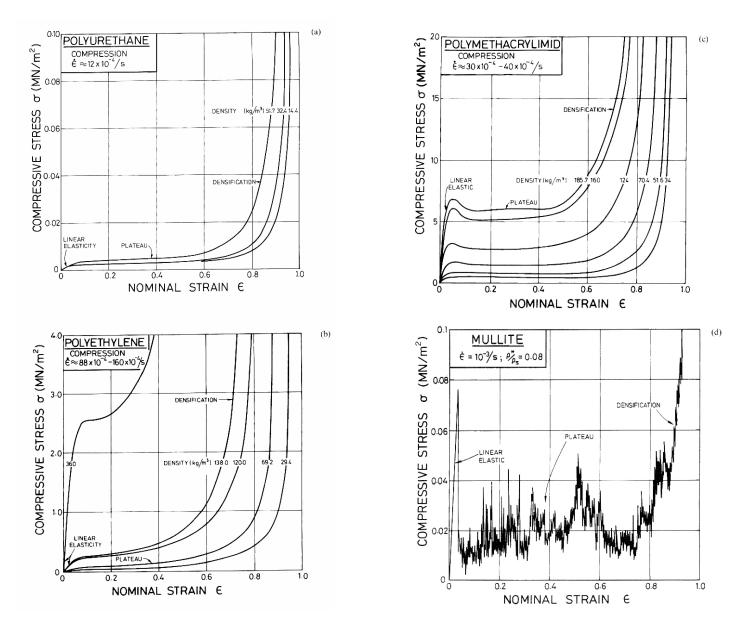
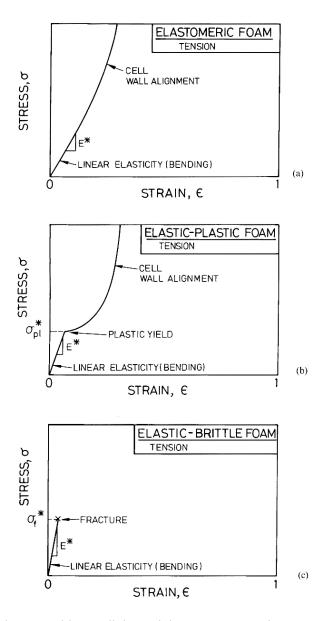
<u>Open-cell foans</u> • stress - strain curves : de formation + failure mechanisons • compression - 3 regimes - linear elastiz - bending - stress platean - cell collapse by buckling yielding crushing - densification - yielding can occur - bittle fracture

Linear elastic behaviour

- · initial linear elasticity bending of celledges (small 4)
- · as the 1, axial deformation becomes more significant
- · consider dimensional argument, which models mechanism of deformation + failure, but not cell geometry
- · consider cubic cell, square coss-section members of area t2, length, l



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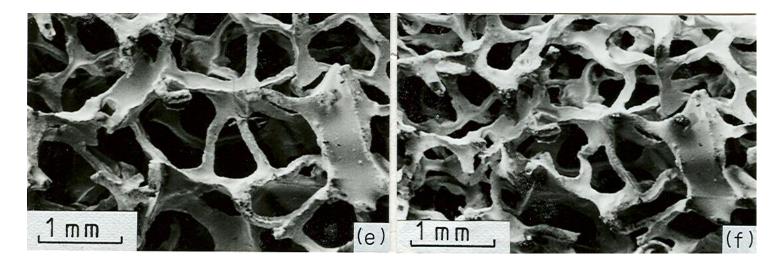


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Foams: Bending, Buckling

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Foams: Plastic Hinges



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Foams: Cell Wall Fracture

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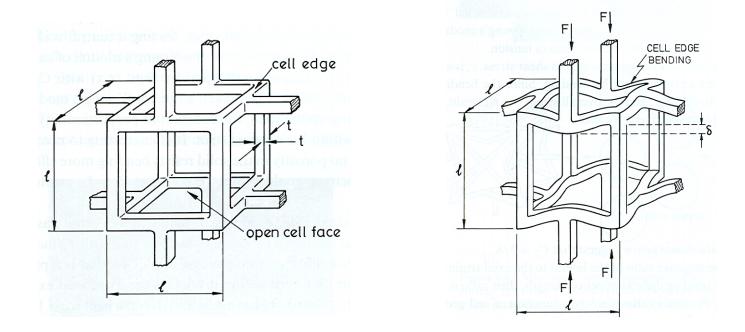
regardless of specific cell geometry chosen: $p^{*}|_{p_{s}} \propto (t_{0})^{2} \qquad I \ll t^{4}$ $\sigma \ll F/l^{2} \qquad E \ll S/l \qquad J \ll Fl^{3}/E_{s}I$ $E^{*} \propto E \propto F l \approx F l \approx F \frac{E_{s}t^{4}}{2} \ll E_{s}(\frac{t}{l})^{4} \ll E_{s}(f_{p_{s}}^{*})^{2}$ $E^{*} = C_{1} E_{s} (p^{*}/p_{s})^{2} \qquad C_{1} \text{ includes all geometrical constants.}$ $Data: C_{1} \approx I$

- · data suggest C1 =1
- analysis of open cell tetrakaideca hedral cells with Platean boides gives C1 = 0.98
- shear modulus $G^* = C_2 E_s (p^*/s)^2$

Poisson's ratio
$$v^* = \frac{E}{2G} - 1 = \frac{C_1}{2C_2} - 1 = Constant, nidependent of
 $v^* = C_3$

(analogous to honey cambs m-plane)$$

Foam: Edge Bending



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Poisson's ratio

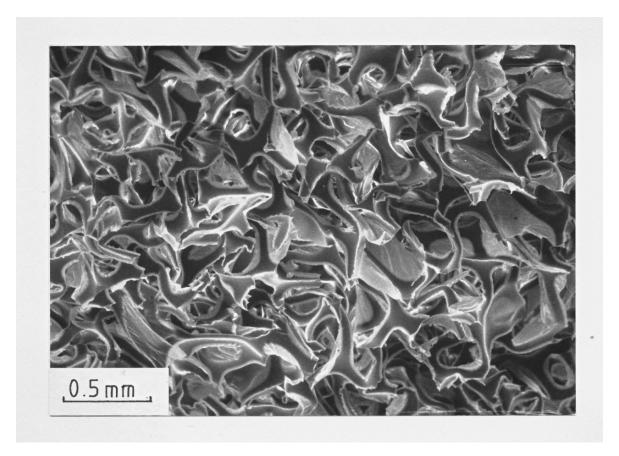
- · can make negative Poisson's ratio forms
- · Mvert cell angles (analogous to honeycamb) ·
- eg. the/moplastic foans load hydrostatically +heat to T7Tg, then cool + release load

so that edges of cell permanently point invaria-

Closed-cell forms

- · edge bending as for open-cell forms
- · also: face stretching + gas compression
- · polymer foans: surface tension drows material to edges during processing · define te, tf in figure
- apply F to the cubic structure

Negative Poisson's Ratio



Gibson, L. J., and M. F. Ashby. *Cellular Solids: Structure and Properties*. 2nd ed. Cambridge University Press, © 1997. Figure courtesy of Lorna Gibson and Cambridge University Press.

- · external work done a FS
- · internal work bending edges & Fe Se & E, I S?
- · inkinal work stretching faces a $\sigma_f \in_f V_f \propto E_s \in_f^2 V_f \propto E_s (\delta_f)^2 t_f l^2$ $\therefore F \delta = \alpha E_s t_e^4 \delta^2 + \beta E_s (\frac{\delta}{2})^2 t_f l^2$ $E^* \propto \frac{F}{\sigma^2} \frac{l}{s} = \rho F \alpha E^* \delta l$

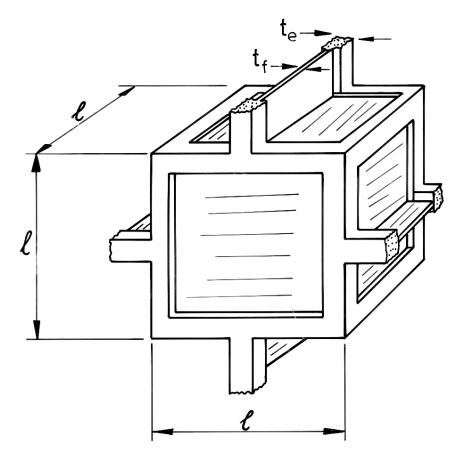
$$\therefore E^* S^{\chi} l = \alpha \frac{E_s te^4}{l^3} S^2 + \beta E_s \left(\frac{s}{k}\right)^2 t_f l^2$$

$$E^{*} = \alpha E_{s} (te/\varrho)^{4} + \beta E_{s} (tf/\varrho)$$
Nok: open cells, uniform t: if $\phi = volume$ fraction of solid in cell edges:

$$p^{*}/p_{s} \propto (t/\varrho)^{2} \qquad te/\varrho = C \phi^{1/2} (\rho^{*}/p_{s})^{1/2}$$

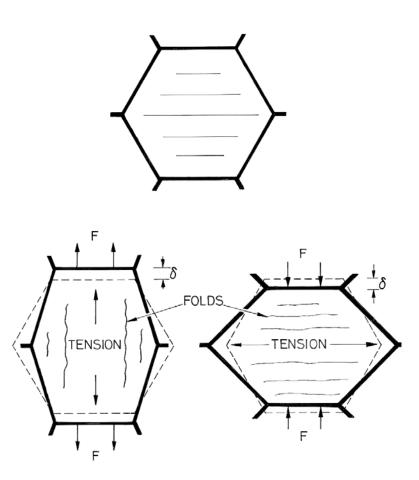
closed cells, uniform t $p^{*}/_{s} \propto (4e)$ $\frac{E^{*}}{E_{s}} = C_{1} \phi^{2}(\frac{p^{*}}{p_{s}})^{2} + C_{1}^{'}(1-\phi) \frac{p^{*}}{p_{s}}$

Closed-Cell Foam



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Cell Membrane Stretching



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Closed cell forms - gas within cell may also contribute to E*

- · Cubic element of form of volume V.
- · under uniaxial stress, axial strain in direction of stress is E
- · deformed volume V is:

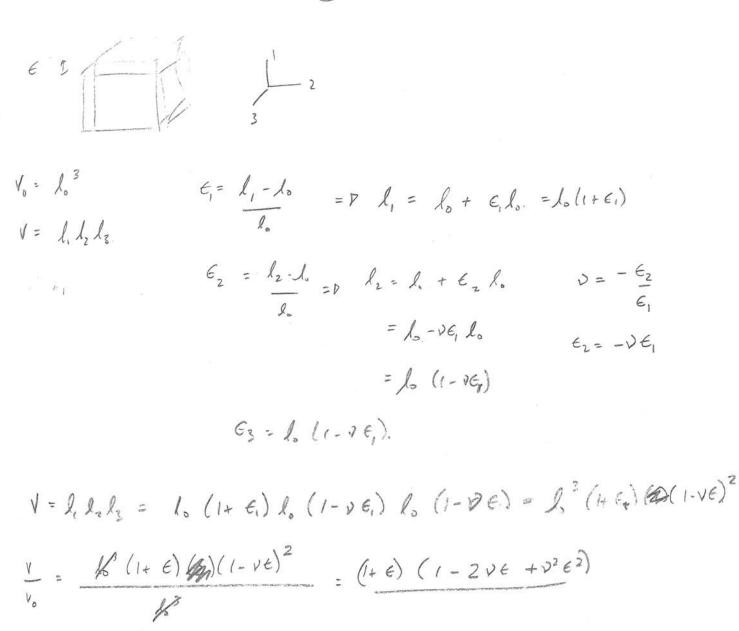
 $\frac{V}{V_{0}} = 1 - E(1 - 2v^{*})$ $\frac{V_{0}}{V_{0}} = \frac{1 - E(1 - 2v^{*})}{1 - p^{*}/p_{s}}$ $\frac{V_{0}}{V_{0}} = \frac{1 - E(1 - 2v^{*})}{1 - p^{*}/p_{s}}$ $\frac{V_{0}}{V_{0}} = \frac{1 - E(1 - 2v^{*})}{1 - p^{*}/p_{s}}$ $\frac{V_{0}}{V_{0}} = \frac{V_{0}}{1 - p^{*}/p_{s}}$ $\frac{V_{0}}{V_{0}} = \frac{v_{0}}{v_{0}}$ $\frac{V_{0}}{v_{0}} = \frac{v_{0}}{v_{0}}$

· Boyle's law: pVg = p.V5

p = pressure after strain E ps = pressure initially (5

- pressure that must be overcome is $p' = p - p_0$ $p' = \frac{p_0 \in (1 - 2v^*)}{1 - \epsilon(1 - 2v^*) - p^* p_s}$

- contribution of gas compression to the modulus, E*: $E_g^* = \frac{dp'}{de} = \frac{p_o(1-2v^*)}{1-p^*/p_s}$



 $= (1 - 2\nu\epsilon + 2\nu\epsilon^{2}) + \epsilon - 2\nu\epsilon^{2} + 2\nu\epsilon^{2}$ $= 14\epsilon \epsilon + 2\nu\epsilon \qquad (4alung comp. cs +)$ $= 1 - \epsilon(1 - 2\nu)$

(49)

$$p' = p - p_{0}$$

$$p = \frac{p_{0}V_{s}^{*}}{V_{s}}$$

$$p' = p - p_{0} = \frac{p_{0}V_{s}^{*}}{V_{s}} - p_{0} = p_{0}\left(\frac{V_{s}^{*}}{V_{s}} - 1\right) + p_{0}U\left(\frac{V_{s}^{*}}{V_{s}} - N_{s}\right)$$

$$= p_{0}\left[\frac{1 - p/s}{1 - \epsilon(1 - 2v^{*}) - p^{*}/s} - 1\right]$$

$$= p_{0}\left[\frac{Y - p/s}{1 - \epsilon(1 - 2v^{*}) - p^{*}/s}\right]$$

$$= p_{0}\left[\frac{\xi(1 - 2v^{*})}{1 - \epsilon(1 - 2v^{*}) - p^{*}/s}\right]$$

(56)

8

Closed cell form

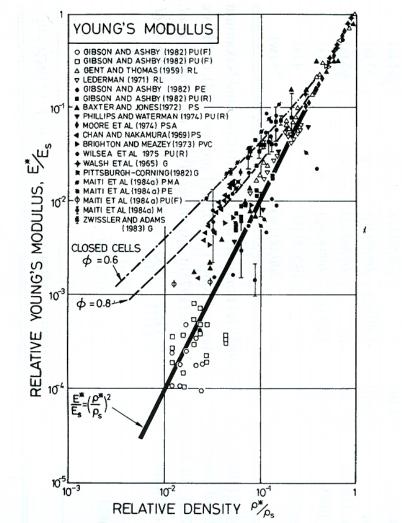
$$\frac{E^*}{E_s} = \oint^2 (f_s^*)^2 + (1-\varphi)(f_s^*) + \underbrace{Po(1-2v^*)}_{E_s} \underbrace{1-p^*(p_s)}_{E_s}$$

edge bending face strething gas compression
· note: If $P_s = Patm = 0.1 MPa$, gas compression term negligible,
except for closed-cell elastometric focus
· gas comp. can be significant if P_s ?? Patm; also modifies shape of stress plateau
in elastometric closed cell focus
Shear Modulus: edge bending, face stretching; shear $\delta V = o$ gas contrib. = o
 $\frac{E^*}{E_s} = \frac{3}{8} \left[\oint^2 (f_s^*)^2 + (1-\theta)(f_s^*) \right]$
Poisson's ratio = $f(cell geometry only)$ $V^* \approx V_3$
Comperison with data

6

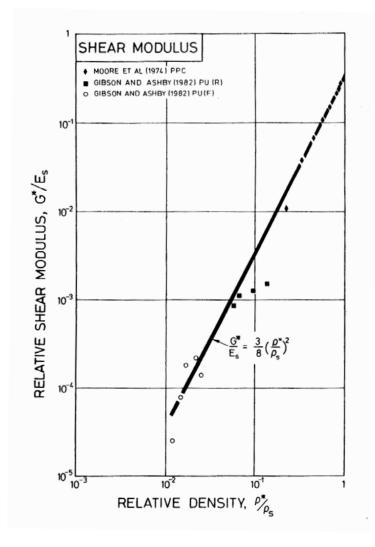
- · data for polymers, glasses, elastomers
- · Es, ps Table 5.1 in book
 - · open cells open symbols · closed cells filled symbols





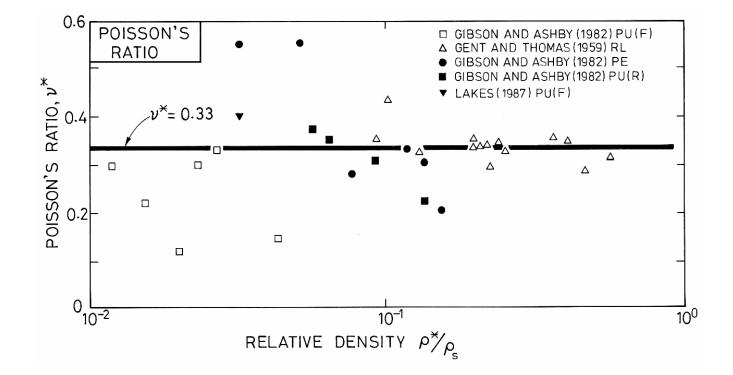
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Shear Modulus



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Poisson's Ratio



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$$\frac{D per cells:}{P_{cr}} = n^{2} \pi^{2} \frac{\pi^{2} E_{s} I}{l^{2}}$$

$$\sigma_{e1}^{*} \propto \frac{P_{cr}}{l^{2}} \propto \frac{E_{s} (\frac{t}{l})^{4}}{l^{2}}$$

$$Data: C_{4} \approx 0.05, \quad collesponds to strain when the buckling mittates, since $E^{*} = E_{s} (e^{*}/s)^{2}$$$

Closed cells

Non-linear elasticity

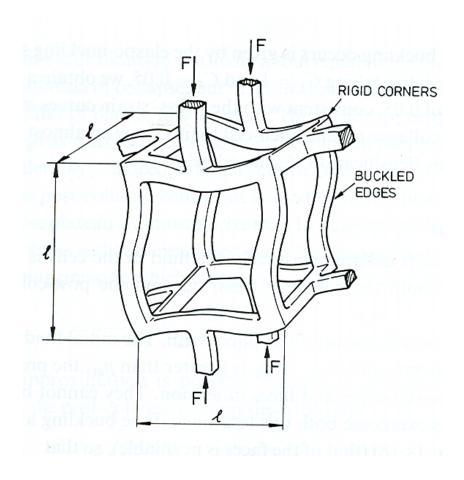
• t_f often small compared to the (surface tension in processing) - contrib. small • $if p_o \gg p_{atm}$, cell walls pretensioned; buckling stress has to overcame this $\delta_{el}^* = 0.05 E_s (\rho^* | \rho_o)^2 + P_o - P_{atm}$

- post-collapse behaviour - stress plateau rises due to gas compression (if faces don't rupture) N = 0 in post-collapse regime

$$P' = \frac{p_{o} \in (1 - 2v^{*})}{1 - \in (1 - 2v^{*}) - p^{*}} p_{s} = \frac{p_{o} \in e}{1 - e - p^{*}} p_{s}$$

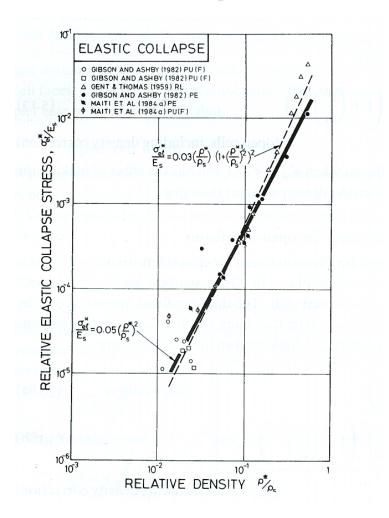
 $\sigma^{*} = 0.05 E_{s} \left(\frac{p}{p_{s}} \right)^{2} + \frac{p_{0} E}{1 - E - p^{*} p_{s}}$

Elastic Collapse Stress



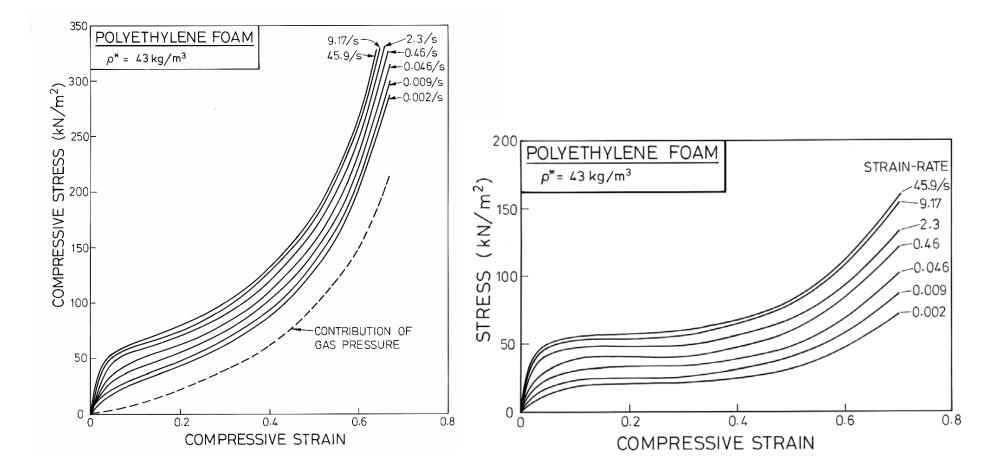
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Elastic Collapse Stress



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Post-collapse stress strain curve



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Plastiz collapse

Open cells

· faiture when M=Mp

$$M_{p} \propto \sigma_{ys} t^{3} \qquad M \propto \sigma_{pl}^{*} l^{3}$$

$$\sigma_{pl}^{*} = C_{s} \sigma_{ys} \left(\rho^{*} |_{\mathcal{B}} \right)^{3/2} \qquad C_{s} \sim 0.3, \text{ from Lata.}$$

· elastic collapse precedes plastic collapse if the < of pe

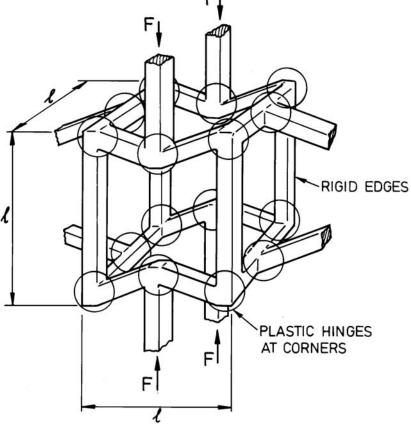
$$0.05 E_{s} (p^{*}|p_{s})^{*} \leq 0.3 \sigma_{y_{s}} (p^{*}|p_{s})^{s_{12}}$$
rigid polymers $(p^{*}|p_{s})_{cr} < 0.04 (\frac{\sigma_{y_{s}}}{E_{s}}, \frac{1}{s_{0}})$

$$(p_{1}|p_{s})_{critical} \leq 36 (\sigma_{y_{s}}|E_{s})^{*}$$

$$Metals (p^{*}|p_{s})_{cr} < 10^{-5} (\frac{\sigma_{y_{s}}}{E_{s}}, \frac{1}{1000})$$

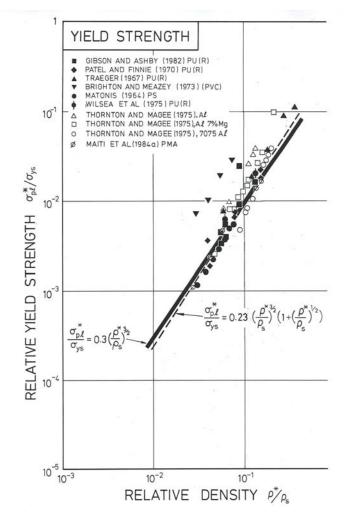
Closed cells

Plastic Collapse Stress



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Plastic Collapse Stress



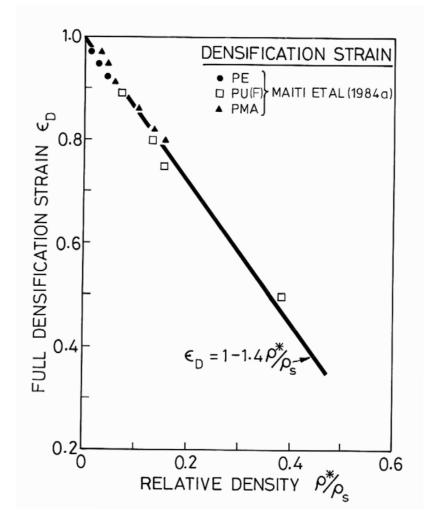
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Srittle crushing strength
Open cells
· failure when
$$M = M_f$$
 $M \propto \sigma_{cr}^* l^2$ $M_f \propto \sigma_{fs} t^3$
 $\sigma_{cr}^* = \zeta_s \sigma_{fs} (p^{*l}p_s)^{3k}$ $C_6 \approx 0.2$

Densification strain, Ep

- · at large comp. shain, cell walls begin to touch, J-E rises steeply
- · E* > Es ; J-E curve looks vertical, at limiting stram
- · Might expect ED = 1- p*/3
- · Valls jan together at slightly smaller strain than this: $E_p = 1 - 1.4 p^{*1}p_s$

Densification Strain

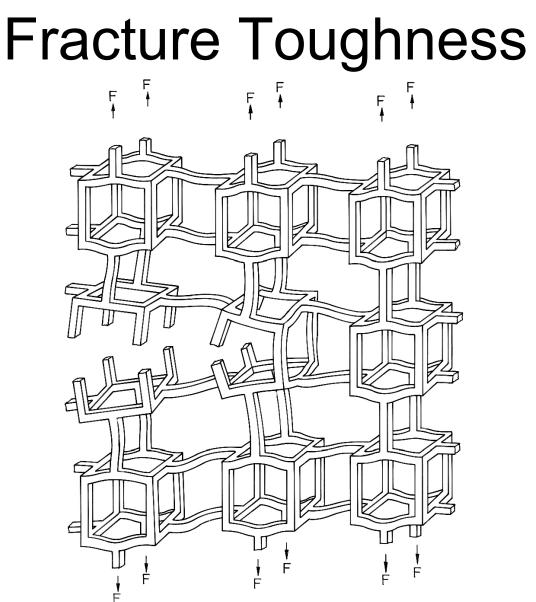


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<u>Fracture toughness</u> Open cells : cract length 2a, local stress σ_{ℓ} , remote stress σ^{∞} $\sigma_{\ell} = C \sigma^{\infty} \sqrt{\pi a}$ a distance r from crack tip $\sqrt{2\pi r}$

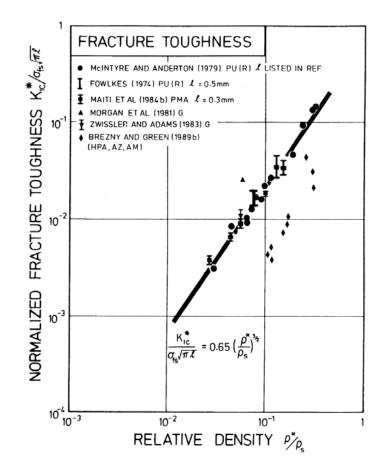
next unbroken cell wall a distance r \$ l/2 a Lead of crack tip subject to a force (integrating stress over next cell)
 F & 5 l² & 0[®] [a l²

• edges fail When applied moment, $M = \text{fractive moment}, M_f$ $M_f \not a \sigma_{fs} t^3$ $M \propto Fl^2 \propto \sigma^{\infty} \left(\frac{G}{Q}\right)^{1/2} l^3$ $M = M_f = P \sigma^{\infty} \left(\frac{G}{Q}\right)^{1/2} l^3 \not a \sigma_{fs} t^3$ $\sigma^{\infty} \propto \sigma_{fs} \left(\frac{L}{Q}\right)^{1/2} \left(\frac{t}{Q}\right)^3$ $K_{Tc}^* = \sigma^{\infty} \sqrt{\pi a} = C_8 \sigma_{fs} \sqrt{\pi l} \left(\frac{\varphi^*}{\rho_s}\right)^{3/2}$ Data: $C_8 \sim 0.65$



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Fracture Toughness



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